

9 - Beam Reflections

ρ = radius of curvature

$$\frac{1}{\rho} = \frac{M(x)}{EI} \approx \frac{d^2 y}{dx^2}$$

Integrate once:

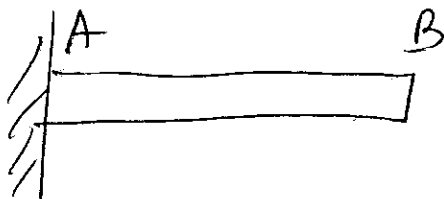
$$\frac{dy}{dx} = \tan \theta \approx \theta(x) \rightarrow \theta(x) = \int \frac{M(x)}{EI} dx + C_1$$

← slope (in radians)

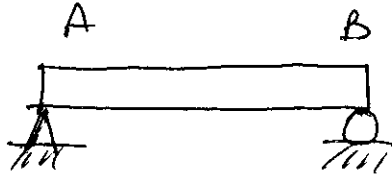
Integrate again:

$$y(x) = \int \theta(x) dx + C_2 = \text{deflection} = \text{elastic curve}$$

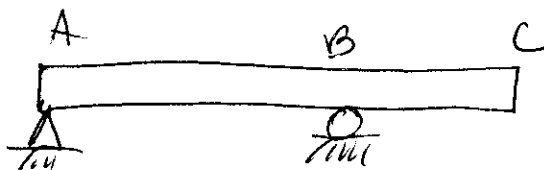
Apply Boundary Conditions to evaluate C_1, C_2 :



$$y_A = 0, \theta_A = 0$$



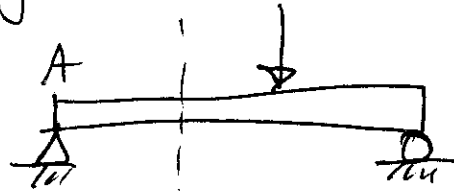
$$y_A = 0, y_B = 0$$



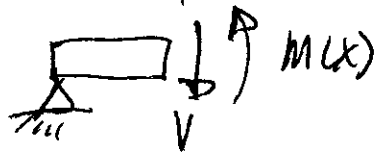
$$y_A = 0, y_B = 0$$

Determine $M(x)$ by slicing through section where you want to know $y(x)$ and

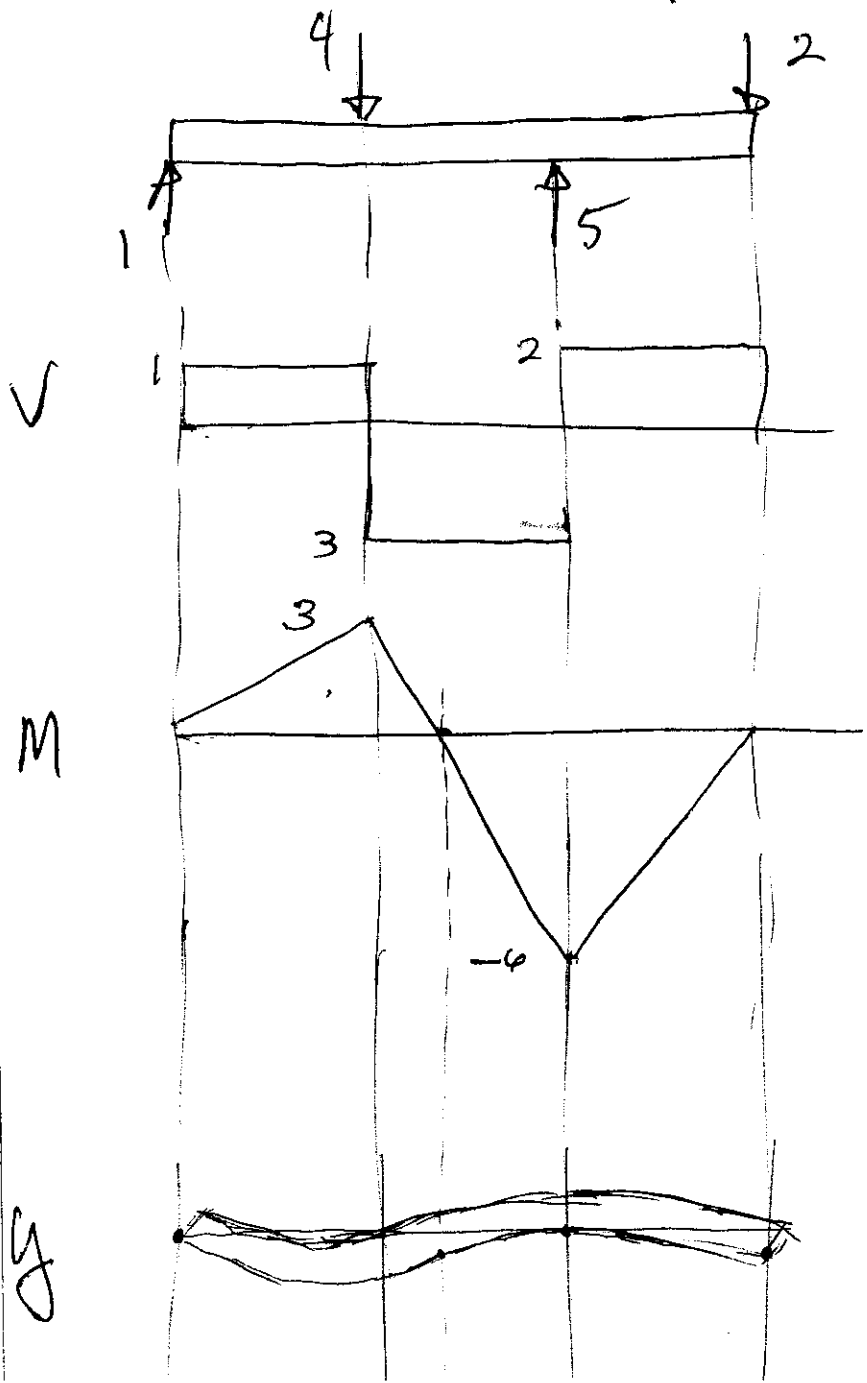
$$\sum M_A = 0$$



$$M(x) = \text{---}$$



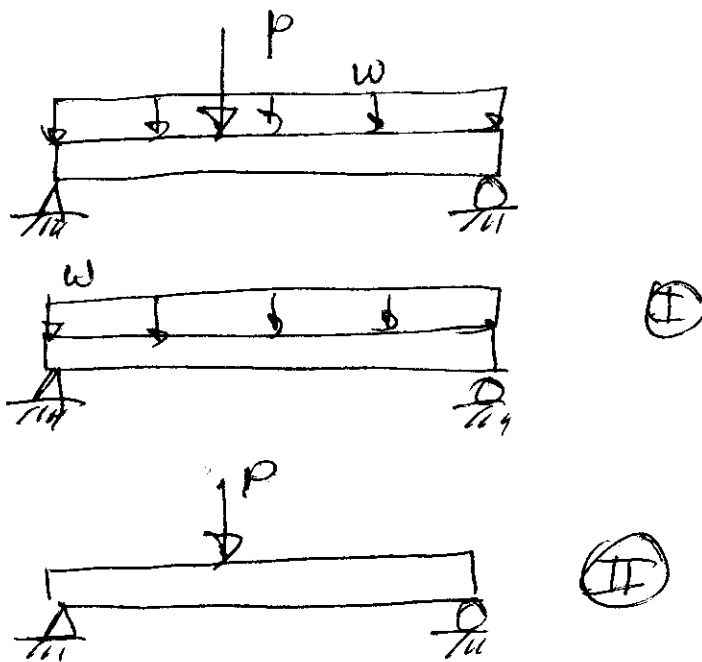
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Superposition:

- Complicated loadings can be decomposed into several simple loads
- the deflection (y) and slope (θ) for the original problem can be obtained by adding the y 's, θ 's from the simple loading cases.
- This is facilitated by the use of tables

Ex:



$$y(x) = [y(x)]_I + [y(x)]_{II}$$

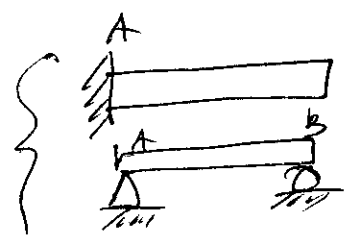
$$\theta(x) = [\theta(x)]_I + [\theta(x)]_{II}$$

look these up in tables.

Statically Indeterminant Problems

Statics: $\sum F_y = 0$, $\sum M = 0$, ~~$\sum F_x = 0$~~

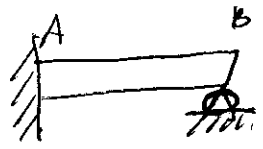
2 eqns
↳ ok for



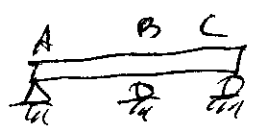
M_A, R_A

R_B

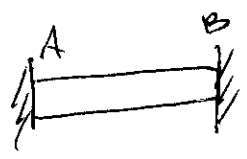
If we have "extra" supports, then beams statically indeterminant:



3 reactions: M_A, R_A, R_B



: R_A, R_B, R_C



4 reactions: M_A, M_B, R_A, R_B

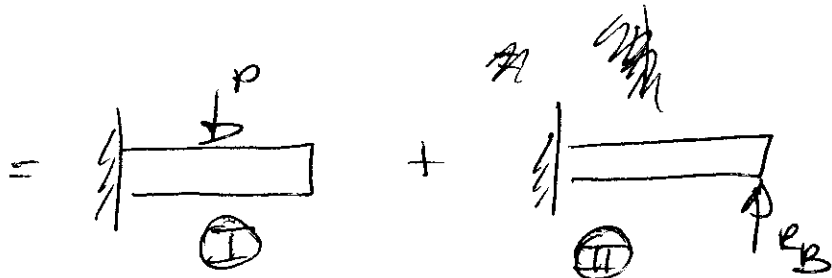
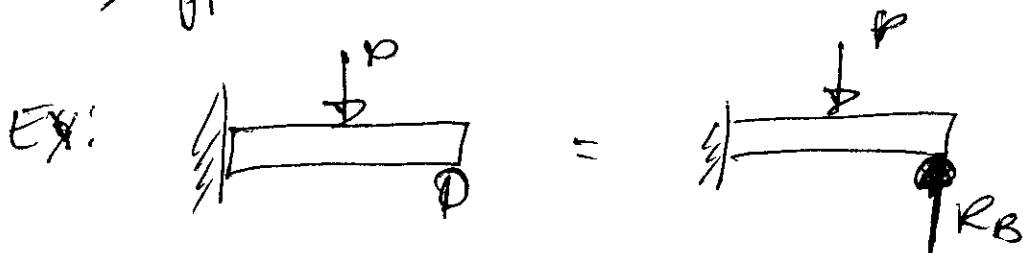
Use deflections, slopes to solve for one or more of reactions, then use statics to find the rest.

Procedure:

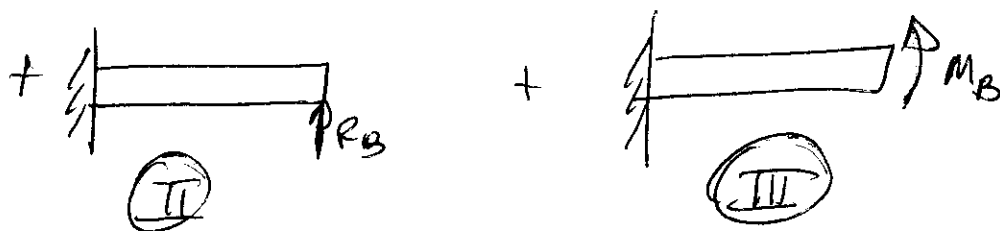
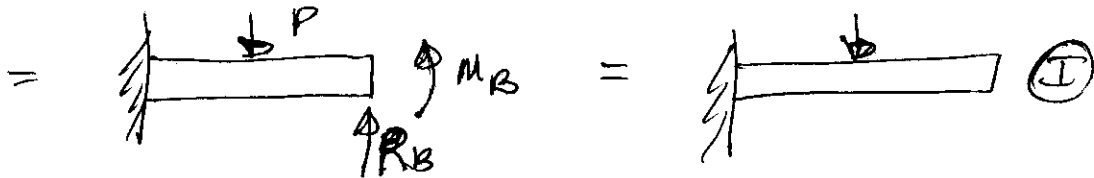
- 1) "Release" reactions one @ a time until beam is statically determinant.

- 2) Replace each released reaction
 - w) a corresponding applied force/moment
- 3) use super position to find y and/or θ @ released support
- 4) set y and/or θ @ released support = 0, solve for unknown reaction(s)

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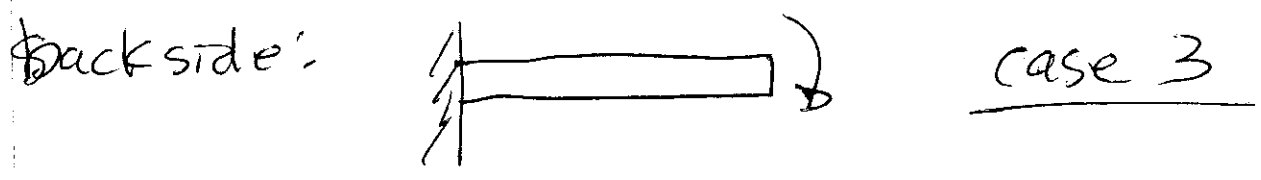
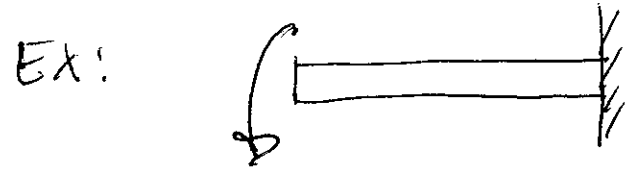
$$y_B = (y_B)_I + (y_B)_{II} = 0 \rightarrow \text{Solve for } R_B$$



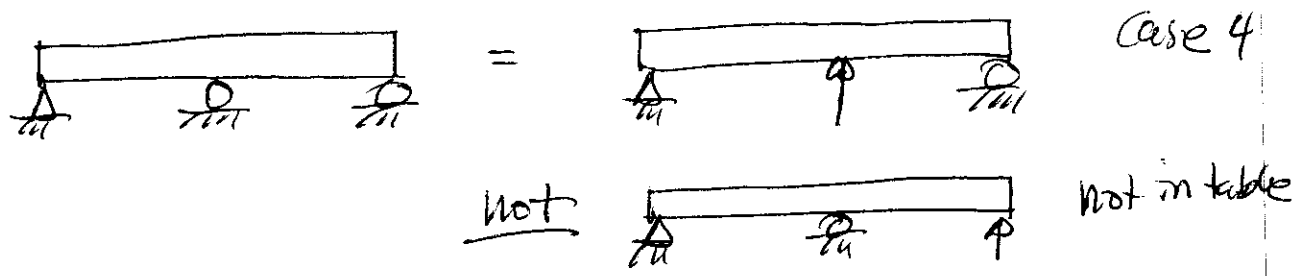
$$\left. \begin{aligned} y_B &= (y_B)_I + (y_B)_II + (y_B)_III = 0 \\ \theta_B &= (\theta_B)_I + (\theta_B)_II + (\theta_B)_III = 0 \end{aligned} \right\} \text{ solve for } R_B, \theta_B$$

Tips for using tables:

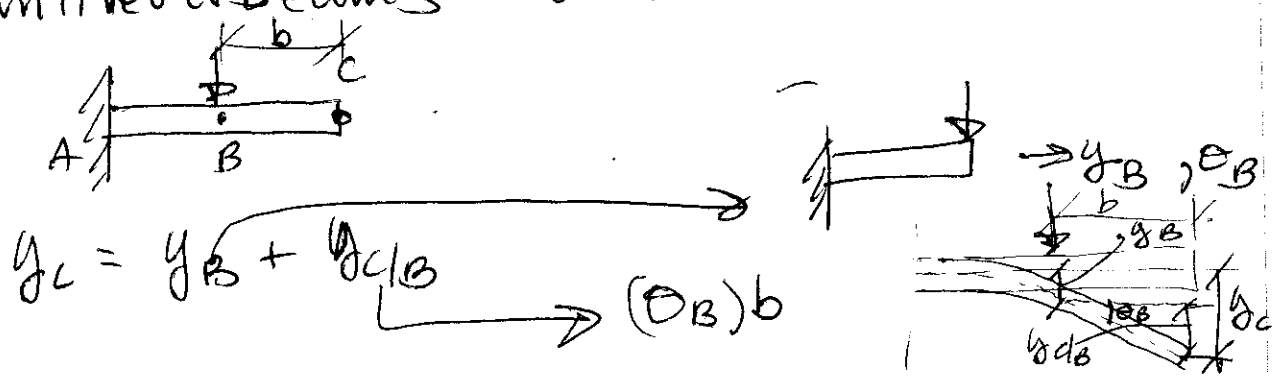
- Sometimes it is convenient to redraw beams looking @ "back side"



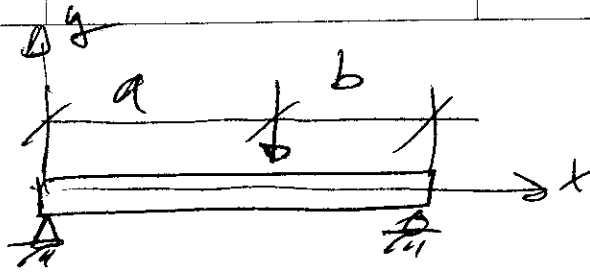
- decompose your beam into pieces that look like the table:



- Cantilever beams loaded in middle:



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Case 5

table gives $y(x) = \text{-----}$ for $x \leq a$

for $y(x)$ for $x > a$:

replace x with $(L-x)$ and swap a, b
In eqns for $y(x)$ for $x < a$

$x < a$
 $\rightarrow y(x) = \frac{Pb}{6EIL} [x^3 - (L^2 - b^2)x]$

$x > a$
 $y(x) = \frac{Pq}{6EIL} [(L-x)^3 - [L^2 - a^2](L-x)]$

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10 - Buckling

Centric Axial loading of perfectly straight + pin-ended column

Elastic Curve: $y(x) = A \sin \lambda x + B \cos \lambda x$

$$\lambda^2 = \frac{P}{EI}$$

Solution to Diff eq: $\frac{d^2 y}{dx^2} + \lambda^2 y = 0$

Apply B.C.s to find A, B

$$x=0 \rightarrow y=0 \rightarrow B=0$$

$$x=L \rightarrow y = A \sin \lambda L = 0$$

two options:

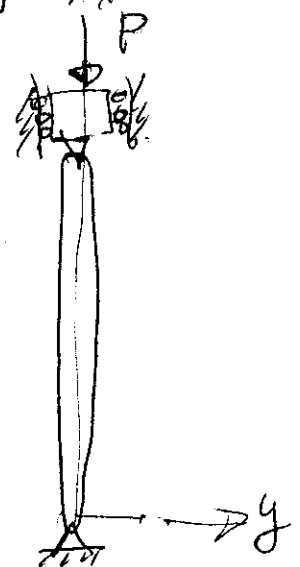
1) $A=0 \rightarrow y(x)=0$ uniform shortening

$$2) \sin \lambda L = 0 \rightarrow \lambda L = \pi n = \sqrt{\frac{P}{EI}}$$

$$P = \frac{\pi^2 n^2 EI}{L} \quad n=1 \text{ (Smallest } P)$$

$$P_{cr} = \frac{\pi^2 EI}{L}$$

$P \geq P_{cr} \rightarrow$ column buckles



Can be extended to other end conditions:

$P_{cr} = \frac{\pi^2 EI}{L_e^2}$ → See overhead of end conditions.

Note:

With Bending - we can choose which way we load the beam - so we can choose bending about strong axis (larger I)

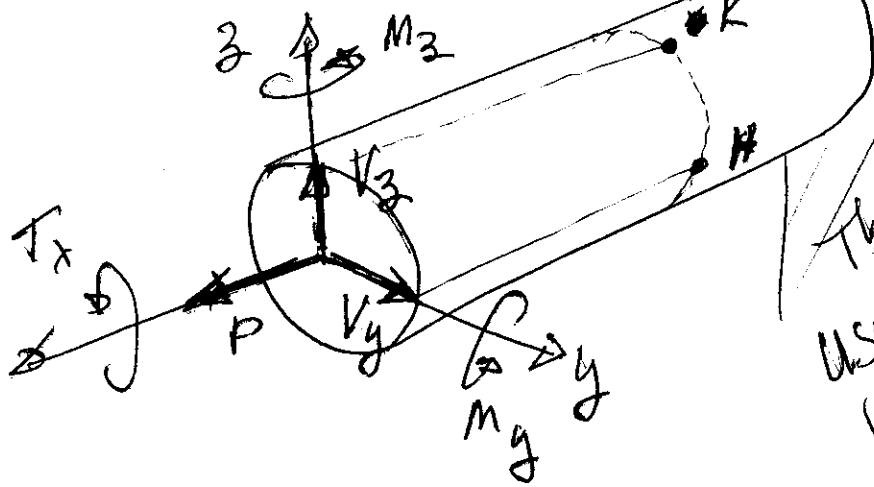
→ for Buckling - columns will buckle about weak axis - so you need to use smaller I ~~I~~ to determine P_{cr} !!

Ways to Increase P_{cr}

- Increase E
- Increase I
- decrease L_e
 - decrease L
 - change supports (i.e. pinned to fixed)



8 - Combined loading



This class:
usually V_z or V_y
but not both

Normal Stresses

$$\sigma_x = \frac{P}{A} - \frac{M_z y}{I_z} + \frac{M_y z}{I_y}$$

Pressure Vessels

$\sigma_1 = \frac{pr_i}{t}$ = hoop stress = circumferential stress

$\sigma_2 = \frac{pr_i}{2t}$ = longitudinal stress = σ_x

→ @ H, $\sigma_1 = \sigma_z$, @ K, $\sigma_1 = \sigma_y$

Shear stresses

$$\tau = \tau_1 + \tau_2$$

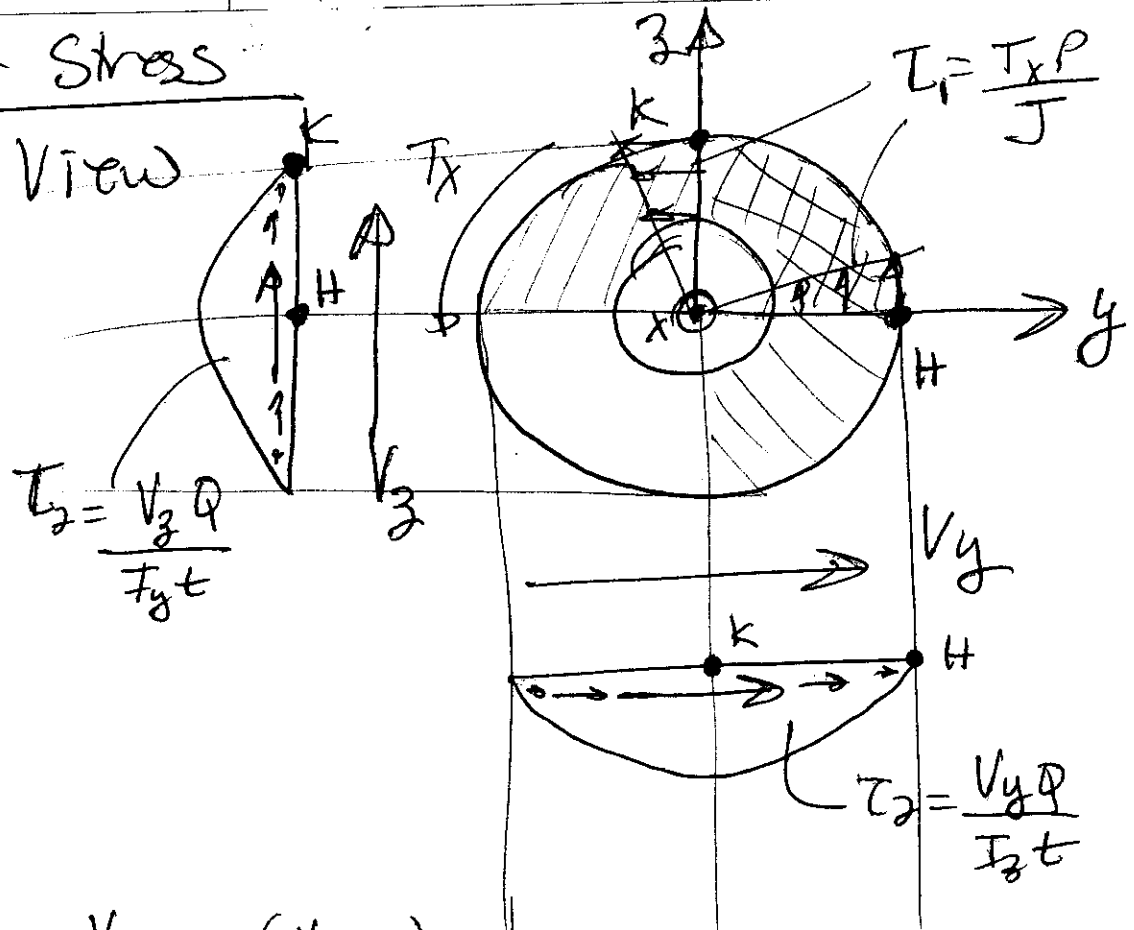
$$\tau_1 = \frac{T_x \rho}{J}, \quad \tau_2 = \frac{V_z \rho}{I_y t} \quad \text{or} \quad \frac{V_y \rho}{I_z t}$$

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Shear Stress

End View



t =

a) T_x, V_z ($V_y = 0$)

K: $\tau_2 = 0, \tau_k = \tau_1 = \frac{T_x r_0}{J}$

H: $\tau_2 = \frac{V_z Q_{max}}{I_y t}, Q_{max} = \frac{2}{3} (r_0^2 - r_i^2)$

$Q = A \bar{y} = \left(\frac{\pi r^2}{2}\right) \left(\frac{4r}{3\pi}\right) = \frac{2}{3} r^2$

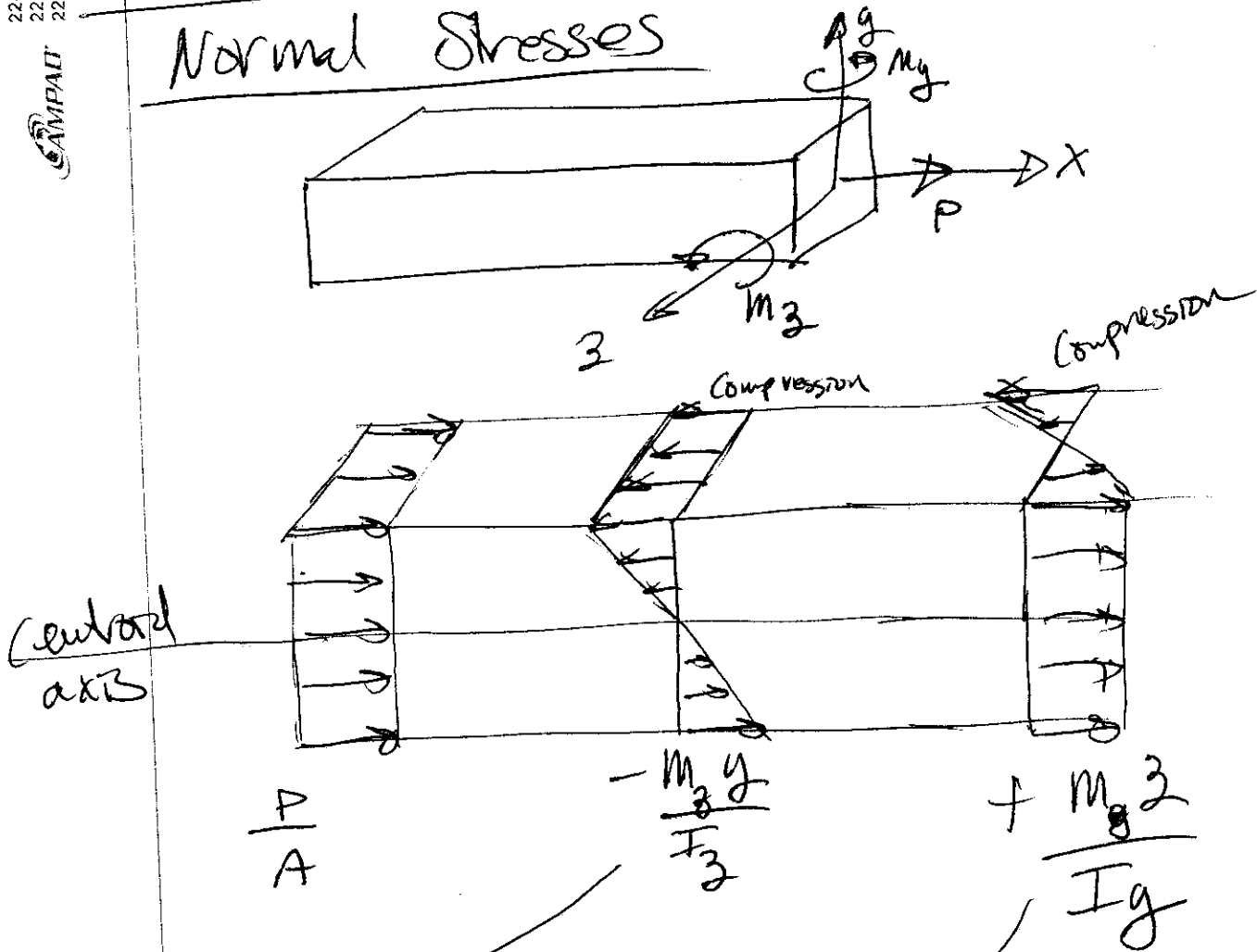
$\tau_H = \frac{T_x r_0}{J} + \frac{V_z Q_{max}}{I_y t}$

b) T_x, V_y ($V_z = 0$)

H: $\tau_z = 0, \tau_x = \tau_y = \frac{T_x r_0}{J}$

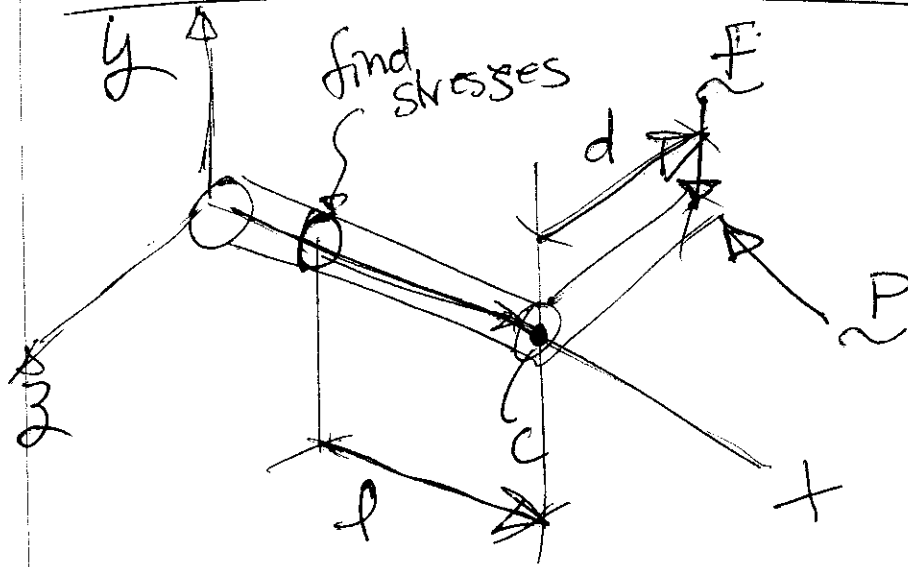
K: $\tau = -\frac{T_x r_0}{J} + \frac{V_y \rho}{I_z t}$

Normal Stresses



$y =$ } distance from
 $z =$ } centroidal axis

Tips for Combined loading



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1) the location where you must find the stresses — this is your "beam" or "rod".

2) Move all the forces, moments to the centroid of the section of the beam
point C

3) Put forces, lengths in vector format;

$$F = -F\hat{j} \quad , \quad P = -P\hat{j}$$

$$d = -d\hat{i} \quad , \quad l = +l\hat{i}$$

Axial load : $\tilde{P} = -P \hat{i}$

Shear : $\tilde{V}_y = -f \hat{j}$

Torsion : $\tilde{T}_x = \underline{d} \times \underline{f} = (-d\hat{i}) \times (-f\hat{j})$
 $= -df \hat{k}$

Bending : $\tilde{M}_y = \underline{d} \times \underline{P} = (-d\hat{i}) \times (-P\hat{i}) = +dP \hat{j}$

$\tilde{M}_z = \underline{l} \times \underline{F} = (l\hat{i}) \times (-f\hat{j}) = -lf \hat{k}$

